Airbus Helicopters Inc.
Technical Support
2701 Forum Drive
Grand Prairie, Tx. 75052

Subject: Global AMOC for A.D. 2007-25-08

To All AS365/EC155 Operators,

On January the 14th 2008 A.D. 2007-25-08 was issued for all AS365’s and EC155’s concerning the tail rotor gear boxes.

The attached AMOC relieves customers from having to comply with A.D. 2007-25-08 if a 365A33-6005-09 or subsequent tail rotor gearbox is installed. These gearboxes have Mod 07-65B63 incorporated. MOD 07-65B63 is a terminating action for ASB 05.00.61 for the 365’s and ASB 05A022 for the 155’s both of these ASB’s are currently at revision 3.

Respectfully,

Mark Jones
Director, Technical Support
Airbus Helicopters Inc.
December 15, 2014

Airbus Helicopters Inc.
Technical Support Group
2701 Forum Drive
Grand Prairie, TX 75052-7099
Attn: Mr. Mark Jones
Director, Technical Support

Dear Mr. Jones,

This is to acknowledge receipt of your letter, dated 24 November 2014, requesting an Alternate Means of Compliance (AMOC) for Airworthiness Directive (AD) 2007-25-08, dated 14 January 2008. This AD applies to all Airbus (formerly Eurocopter France) Helicopters, model SA-365N1, AS-365N2, AS-365N3, SA-366G1, EC 155B, and EC155B1 helicopters, with a tail rotor (T/R) pitch control rod assembly double bearing installed, certificated in any category.

Your AMOC proposal requests relief for the applicable helicopter models of FAA AD 2007-25-08 that have been modified in accordance with modification (MOD) AMS 07-65B63 OR installation of tail gear box part number (P/N) 365A33-6005-09. You further state that the modification and tail gear box P/N were outlined in Airbus Alert Service Bulletins (ASB’s) 05.00.61 for the AS365 and ASB 05A022 for the EC155B, both at revision 3. You also state that these ASB’s were terminating action for EASA AD 2011-0145, which caused EASA to supersede to EASA AD 2012-0170R1 (EASA approval number 10040799 REV. 1, dated 21 August 2012).

Your AMOC is approved. This is based on EASA AD 2012-0170R1, which states that: TGB part number 365A33-6005-09(MOD 07 65B63) are not affected by degradation of TGB control rod bearing. Your AMOC proposal is acceptable for paragraphs (a), (b), (c), (d), (e), and (h) of FAA AD 2007-25-08, dated 14 January 2008.

Note that the granting of this AMOC does not relieve, alter, or waive any other maintenance operations required for these helicopter models, other than those specifically mentioned herein.

This FAA AMOC is transferable with the aircraft to an operator who operates the aircraft under U.S. registry.

Before using this AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.
All provisions of AD 2007-25-08 that are not specifically referenced above remain fully applicable and must be complied with accordingly.

The Rotorcraft Standards Staff will revoke this AMOC if the Rotorcraft Standards Staff later determines that this AMOC does not provide an acceptable level of safety.

If you have any questions regarding the above information, you may contact Eric Haight at (817) 222-5204.

Sincerely,

[Signature]
James A. Grigg
Manager, Safety Management Group
Rotorcraft Directorate

cc: Mr. Chris Carter, Airbus Helicopters
Good Afternoon,


Should you have any questions, please contact Technical Support at (800) 232-0323.

Thank you,

AIRBUS HELICOPTERS

Kathy Conner
Data Entry Specialist/Tipi Administrator
Airbus Helicopters, Inc.
Office: 972.641.3467
Fax: 972.641.3710
Email: kathy.conner@airbus.com
Good Afternoon,


Should you have any questions, please contact Technical Support at (800) 232-0323.

Thank you,

AIRBUS HELICOPTERS

Kathy Conner
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